Orbit Response Analysis

The method:

- -A small kick $\Delta\theta$ is applied in either the horizontal or vertical
- -Change in closed orbit is measured at each BPM
- -Results summarized in matrix form:

$$\Delta x = M \Delta \theta$$

-Orbit response matrix M has dimensions (horz. + vert. steering eles) x (horiz. + vert. BPMs)

Project outline:

- -Simulate the "ideal" Orbit Response Matrix for current CESR lattice
- -Introduce BPM misalignments / resolution limits into simulated ORM
- -Benchmark: load both matrices into LOCO to find the "misaligned" BPM
- -Analyze the ORM data previously collected, using LOCO
- -Implement BPM corrections suggested by LOCO?

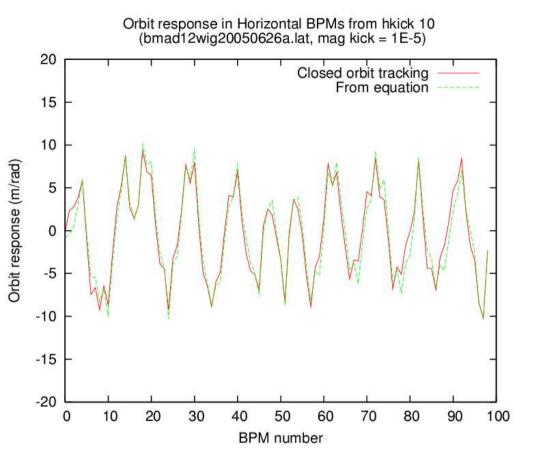
Simulated Ideal ORM

- -Ideal ORM can be calculated using two methods:
 - 1) Directly from the twiss parameters and phase, using $Mij = \sqrt{(\beta i \beta j) * \cos (\pi v |\phi i \phi j|) / [2 \sin (\pi v)]}$ (ith bpm, jth steering element)
 - 2) Manually apply a kick, and measure the difference in closed orbit:

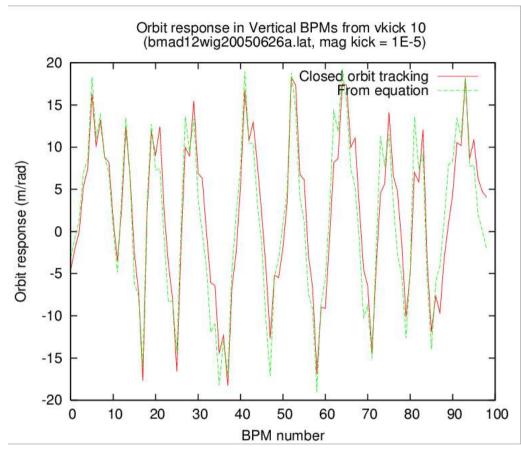
$$Mij = \Delta x/\Delta \theta$$

- -Both methods have been implemented in parallel, allowing us to compare results
- -Results decomposed into four quadrants of the ORM: XX, XY, YX, YY
- -Ideally, XY, YX terms should all be zero (no coupling)
 - -Cannot compute these terms using the first method; only with closed-orbit method

Simulated Ideal ORM – Sample Results

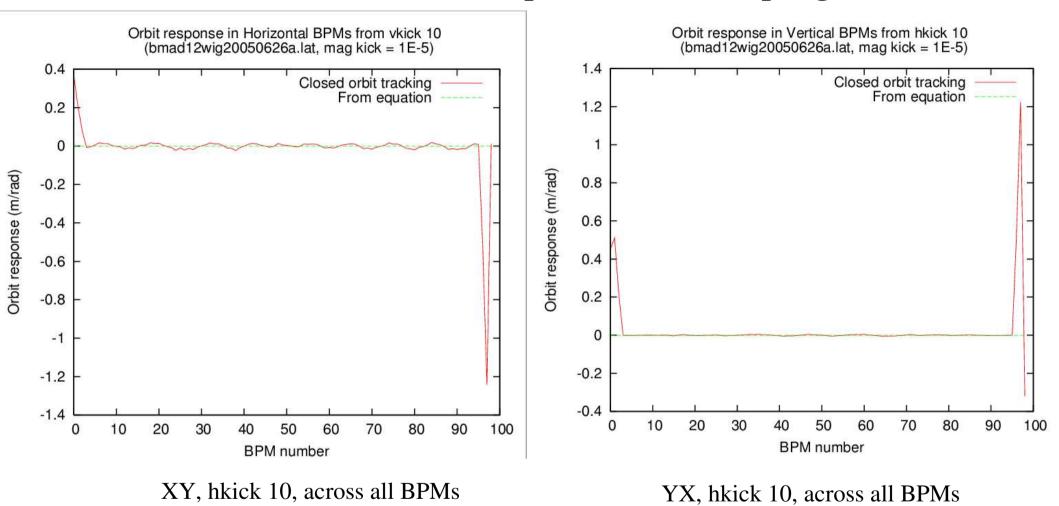


XX, hkick 10, across all BPMs



YY, hkick 10, across all BPMs

Simulated Ideal ORM – Sample Results – Coupling Terms



- -The source of the highly non-zero coupling terms from closed-orbit tracking is possibly due to elements in or near the IR

 -Next step:
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 - -Determine source of these discrepancies
 - -Either introduce BPM rotations and gain errors, or begin analyzing previous ORM data